Computational Fluid Dynamics: Principles and Applications

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The following corrections apply to the 3rd edition of the book published in March 2015 by Elsevier Ltd. (ISBN: 978-0-08-099995-1).

Section 3.1.5: Central and Upwind Schemes

Text on page 42, initial paragraph:

... In this context, it is convenient to differentiate between the discretization of the convective and the viscous fluxes (\vec{F}_c and \vec{F}_v in Eq. (2.19), respectively). ...

Chapter 4: Structured Finite-Volume Schemes

Equation (4.64) on page 96:

$$M_R^- = \left\{ egin{array}{ll} 0 & ext{if } M_R \geq +1 \ -rac{1}{4}(M_R-1)^2 & ext{if } |M_R| < 1 \ M_R & ext{if } M_R < -1 \ . \end{array}
ight.$$

Chapter 5: Unstructured Finite-Volume Schemes

Equation (5.9) on page 128:

$$ec{r}_c = rac{\Omega_{123} \, ec{r}_{c,123} + \Omega_{134} \, ec{r}_{c,134}}{\Omega_{123} + \Omega_{134}} \, .$$

Equation (5.23) and text following on page 135:

$$\vec{n}_{01}\Delta S_{01} = \vec{n}_L \Delta S_L + \vec{n}_R \Delta S_R \,,$$

and the total face area is given by: $\Delta S_{01} = ||\vec{n}_L \Delta S_L + \vec{n}_R \Delta S_R||_2$.

Equation (5.62) on page 153:

$$egin{align} lpha_{ij,1} &= rac{\Delta x_{ij}}{r_{11}^2} \ lpha_{ij,2} &= rac{1}{r_{22}^2} \left(\Delta y_{ij} - rac{r_{12}}{r_{11}} \Delta x_{ij}
ight) \ lpha_{ij,3} &= rac{1}{r_{33}^2} \left(\Delta z_{ij} - rac{r_{23}}{r_{22}} \Delta y_{ij} + eta \Delta x_{ij}
ight) \; , \end{array}$$

Equation (5.74) on page 162:

$$\left(\frac{\partial U}{\partial \ell}\right)_{IJ} \approx \frac{U_J - U_I}{\ell_{IJ}},$$

Chapter 6: Temporal Discretization

Equation (6.85) on page 204:

$$\begin{split} \vec{W}_I^{(k)} &= \vec{W}_I^{(0)} - \frac{\alpha_k \Delta t_I^*}{\Omega_I^{n+1}} \left[\bar{I} + \frac{3}{2 \Delta t} \alpha_k \Delta t_I^* \bar{M}^{n+1} \right]^{-1} \\ &\cdot \left[\vec{R}_I (\vec{W}^{(k-1)}) + \frac{3}{2 \Delta t} (\Omega \bar{M})_I^{n+1} \vec{W}_I^{(k-1)} - \vec{Q}_I^* \right] \,. \end{split}$$

Equations (6.76), (6.77) and the related text on page 199:

$$\vec{R}_I^{(p+1)} pprox \vec{R}_I^{(p)} + \bar{J}_I \Delta \vec{W}^{(p)}$$

with $\Delta \vec{W}^{(p)} = \vec{W}^{(p+1)} - \vec{W}^{(p)}$ and $\vec{J} = \partial \vec{R} / \partial \vec{W}$ being the Jacobian matrix. Inserted into Eq. (6.75) while formulated as an iterative procedure, we obtain for the solution at stage k the expression

$$\left[rac{(\Omegaar{M})_I}{\Delta t} + a_{kk}ar{J}_I
ight]\Deltaec{W}^{(p)} = rac{(\Omegaar{M})_I}{\Delta t}\left(ec{W}_I^n - ec{W}_I^{(p)}
ight) - \sum_{l=1}^{k-1} a_{kl}\,ec{R}_I^{(l)} - a_{kk}\,ec{R}_I^{(p)}\,.$$

This is formally equivalent to Eq. (6.28) and can be solved by any of the implicit methods from the previous subsections. Upon convergence, the solution $\vec{W}^{(p+1)}$ of Eq. (6.77) approximates the intermediate solution at stage k from Eq. (6.75).

Chapter 7: Turbulence Modeling

Equation (7.4) on page 216:

$$\Omega_{ij} = rac{1}{2} \left(rac{\partial v_j}{\partial x_i} - rac{\partial v_i}{\partial x_j}
ight) \, .$$

Equation (7.38) on page 226 and the text below it:

$$\tilde{S} = S + \frac{\tilde{v}}{\kappa^2 d^2} f_{v2} ,$$

$$f_{v1} = \frac{\chi^3}{\chi^3 + C_{v1}^3} , \quad f_{v2} = 1 - \frac{\chi}{1 + \chi f_{v1}} ,$$

$$\chi = \frac{\tilde{v}}{v_I} ,$$

where S stands for the magnitude of the mean rotation rate, that is,

$$S = \sqrt{2 \Omega_{ij} \Omega_{ij}}$$
,

and where Ω_{ij} is given by Eq. (7.4). In order to avoid numerical difficulties, the term \tilde{S} must never become zero or negative. One possibility is simple limiting like $\tilde{S} = \max(\tilde{S}, 0.3 \, S)$. More elaborate approaches

were suggested in Ref. [49].

Equation (7.39) on page 226:

$$f_w = g \left(\frac{1 + C_{w3}^6}{g^6 + C_{w3}^6} \right)^{1/6} ,$$

$$g = r + C_{w2}(r^6 - r) , \quad r = \min \left[\frac{\tilde{v}}{\tilde{S} \kappa^2 d^2}, 10 \right] .$$

Equation (7.41) on page 227:

$$C_{b1} = 0.1355$$
, $C_{b2} = 0.622$, $C_{v1} = 7.1$, $\sigma = 2/3$, $\kappa = 0.41$, $C_{w1} = C_{b1}/\kappa^2 + (1 + C_{b2})/\sigma$, $C_{w2} = 0.3$, $C_{w3} = 2$, $C_{t1} = 1$, $C_{t2} = 2$, $C_{t3} = 1.2$, $C_{t4} = 0.5$.

Text above Equation (7.42) on page 227:

As pointed out in Ref. [47], it is convenient to substitute the diffusion term in Eq. (7.36), that is,

$$\frac{1}{\sigma} \left\{ \frac{\partial}{\partial x_j} \left[(v_L + \tilde{v}) \frac{\partial \tilde{v}}{\partial x_j} \right] + C_{b2} \frac{\partial \tilde{v}}{\partial x_j} \frac{\partial \tilde{v}}{\partial x_j} \right\}$$

by the following expression ...

Text to **Initial and boundary conditions** on page 228:

The initial value of \tilde{v} is usually taken as $\tilde{v}=0.1\,v_L$. The same value is also specified at inflow or far-field boundaries. However, in cases where the transition term f_{t1} (and then mostly also f_{t2}) is omitted, that is, for fully turbulent flows, \tilde{v} should be set to 3–5 times the value of v_L at the inlet or far-field. At outflow boundaries, \tilde{v} is simply extrapolated from the interior of the computational domain. At solid walls, it is appropriate to set $\tilde{v}=0$ and hence $\mu_T=0$.

References on page 246:

[49] Allmaras, S.R.; Johnson, F.T.; Spalart, P.R.: *Modifications and Clarifications for the Implementation of the Spalart-Allmaras Turbulence Model*. ICCFD7-1902, 7th Int. Conf. on Comput. Fluid Dynamics, 2012.

Chapter 8: Boundary Conditions

Equation (8.7) on page 257:

$$(\vec{F}_{c,w} \Delta S)_{i,2} = egin{bmatrix} 0 \ (n_x)_{i-1,2} \ (p_w)_{i-1/4} \ (n_y)_{i-1,2} \ (p_w)_{i-1/4} \ (n_z)_{i-1,2} \ (p_w)_{i-1/4} \ 0 \end{bmatrix} rac{\Delta S_{i-1,2}}{2} + egin{bmatrix} 0 \ (n_x)_{i,2} \ (p_w)_{i+1/4} \ (n_y)_{i,2} \ (p_w)_{i+1/4} \ (n_z)_{i,2} \ (p_w)_{i+1/4} \ 0 \end{bmatrix} rac{\Delta S_{i,2}}{2} \ .$$

Equation (8.8) and related text on page 258:

The pressures $(p_w)_{i-1/4}$ and $(p_w)_{i+1/4}$ in Eq. (8.7) can be obtained by linear interpolation, for example,

$$(p_w)_{i+1/4} = \frac{1}{4}(3p_{i,2} + p_{i+1,2}).$$

Chapter 9: Acceleration Techniques

Point 5 of the solution steps described on page 311:

5. Multiply the residual by $\alpha_k \Delta t/\Omega$.

Equation (9.59) on page 313 (last term in the first column):

$$\bar{P} = \begin{bmatrix} \rho_p & 0 & 0 & 0 & \rho_T \\ \rho_p u & \rho & 0 & 0 & \rho_T u \\ \rho_p v & 0 & \rho & 0 & \rho_T v \\ \rho_p w & 0 & \rho & \rho & \rho_T w \\ \rho_p H + \rho h_p - 1 & \rho u & \rho v & \rho w & \rho_T H + \rho h_T \end{bmatrix}.$$

Text on page 318:

... the condition number Eq. (9.40) is reduced (the condition number is $C_N \approx 1.62$) and the ...